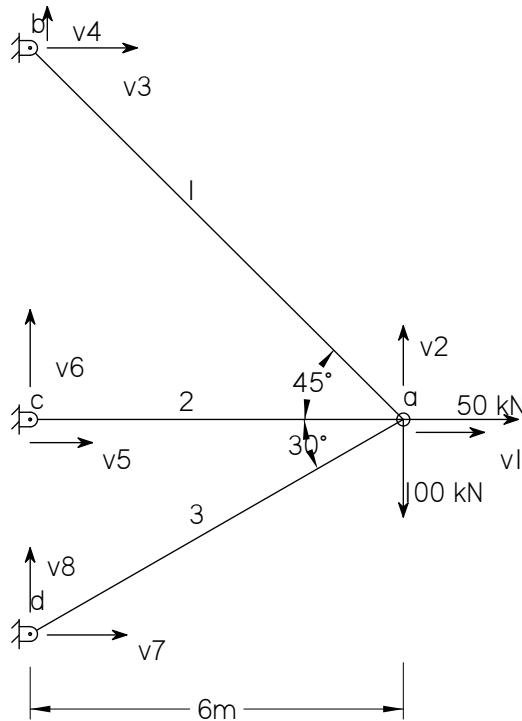


CE 342 - Truss problem

$$C := K \quad kN := 1000 \cdot \text{newton} \quad GPa := 10^9 \cdot \frac{\text{newton}}{m^2}$$

Find the nodal displacements, support reactions, and member forces for the structure shown below.



ne := 3 Number of elements
i := 1..ne
DOFn := 2 Number of degrees of freedom per node
Nn := 4 Number of nodes in system
DOF := Nn·DOFn Total number of DOF for system DOF = 8

Element Properties:

$E_1 := 200 \cdot GPa$ Modulus of Elasticity for all members, kN/mm²

Area of members, m²

$$A_1 := 30 \cdot 10^{-4} \cdot m^2$$

$$A_2 := 20 \cdot 10^{-4} \cdot m^2$$

$$A_3 := 15 \cdot 10^{-4} \cdot m^2$$

thermal coefficients

$$\alpha_1 := 1.2 \cdot \frac{10^{-5}}{C}$$

$$\Delta T := 45 \cdot C$$

Nodal coordinates

$$x := \begin{pmatrix} 6 \\ 0 \\ 0 \\ 0 \end{pmatrix} \cdot \text{m} \quad y := \begin{pmatrix} 6 \cdot \tan(30 \cdot \text{deg}) \\ 6 \cdot \tan(30 \cdot \text{deg}) + 6 \cdot \tan(45 \cdot \text{deg}) \\ 6 \cdot \tan(30 \cdot \text{deg}) \\ 0 \end{pmatrix} \cdot \text{m}$$

Member connectivity

Specify i and j nodes for each member.

$$e_{m1} := \begin{pmatrix} 1 \\ 2 \end{pmatrix} \quad e_{m2} := \begin{pmatrix} 1 \\ 3 \end{pmatrix} \quad e_{m3} := \begin{pmatrix} 1 \\ 4 \end{pmatrix}$$

Find direction cosines

X vector is from end i to end j

$$X_i := \begin{bmatrix} x[(e_i)_2] - x[(e_i)_1] \\ y[(e_i)_2] - y[(e_i)_1] \end{bmatrix} \quad X_3 = \begin{pmatrix} -6 \\ -3.464 \end{pmatrix} \text{m} \quad L_{m3} := |X_i|$$

Direction cosines

cosine

sine

$$c_{m3} := \frac{(X_i)_1}{L_i}$$

$$s_{m3} := \frac{(X_i)_2}{L_i}$$

$$T_{m3} := \begin{pmatrix} c_i & s_i & 0 & 0 \\ 0 & 0 & c_i & s_i \end{pmatrix} \quad T_1 = \begin{pmatrix} -0.707 & 0.707 & 0 & 0 \\ 0 & 0 & -0.707 & 0.707 \end{pmatrix}$$

Member stiffness matrix

Beam members

$$k_i := \frac{A_i \cdot E_i}{L_i} \cdot \begin{pmatrix} 1 & -1 \\ -1 & 1 \end{pmatrix} \quad k_1 = \begin{pmatrix} 70710.68 & -70710.68 \\ -70710.68 & 70710.68 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

$$k_2 = \begin{pmatrix} 6.667 \times 10^4 & -6.667 \times 10^4 \\ -6.667 \times 10^4 & 6.667 \times 10^4 \end{pmatrix} \frac{\text{kN}}{\text{m}} \quad k_3 = \begin{pmatrix} 4.33 \times 10^4 & -4.33 \times 10^4 \\ -4.33 \times 10^4 & 4.33 \times 10^4 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

Calculate element stiffness in global coordinates.

$$K_{m3} := T_i^T \cdot k_i \cdot T_i \quad K_1 = \begin{pmatrix} 35355.3 & -35355.3 & -35355.3 & 35355.3 \\ -35355.3 & 35355.3 & 35355.3 & -35355.3 \\ -35355.3 & 35355.3 & 35355.3 & -35355.3 \\ 35355.3 & -35355.3 & -35355.3 & 35355.3 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

$$K_2 = \begin{pmatrix} 66666.7 & 0 & -66666.7 & 0 \\ 0 & 0 & 0 & 0 \\ -66666.7 & 0 & 66666.7 & 0 \\ 0 & 0 & 0 & 0 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

$$K_3 = \begin{pmatrix} 32476 & 18750 & -32476 & -18750 \\ 18750 & 10825.3 & -18750 & -10825.3 \\ -32476 & -18750 & 32476 & 18750 \\ -18750 & -10825.3 & 18750 & 10825.3 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

Zero out the global stiffness matrix.

$$S := \begin{cases} \text{for } i \in 1 \dots \text{DOF} \\ \quad \text{for } j \in 1 \dots \text{DOF} \\ \quad \quad k_{\text{init},i,j} \leftarrow 0 \\ \quad k_{\text{init}} \end{cases} \quad S = \begin{pmatrix} 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \\ 0 & 0 & 0 & 0 & 0 & 0 & 0 & 0 \end{pmatrix}$$

Assemble global structural stiffness matrix.

$$S := \begin{bmatrix} (K_1)_{1,1} + (K_2)_{1,1} + (K_3)_{1,1} & (K_1)_{1,2} + (K_2)_{1,2} + (K_3)_{1,2} & (K_1)_{1,3} & (K_1)_{1,4} & (K_2)_{1,3} & (K_2)_{1,4} & (K_3)_{1,3} & (K_3)_{1,4} \\ (K_1)_{2,1} + (K_2)_{2,1} + (K_3)_{2,1} & (K_1)_{2,2} + (K_2)_{2,2} + (K_3)_{2,2} & (K_1)_{2,3} & (K_1)_{2,4} & (K_2)_{2,3} & (K_2)_{2,4} & (K_3)_{2,3} & (K_3)_{2,4} \\ (K_1)_{3,1} & (K_1)_{3,2} & (K_1)_{3,3} & (K_1)_{3,4} & 0 & 0 & 0 & 0 \\ (K_1)_{4,1} & (K_1)_{4,2} & (K_1)_{4,3} & (K_1)_{4,4} & 0 & 0 & 0 & 0 \\ (K_2)_{3,1} & (K_2)_{3,2} & 0 & 0 & (K_2)_{3,3} & (K_2)_{3,4} & 0 & 0 \\ (K_2)_{4,1} & (K_2)_{4,2} & 0 & 0 & (K_2)_{4,3} & (K_2)_{4,4} & 0 & 0 \\ (K_3)_{3,1} & (K_3)_{3,2} & 0 & 0 & 0 & 0 & (K_3)_{3,3} & (K_3)_{3,4} \\ (K_3)_{4,1} & (K_3)_{4,2} & 0 & 0 & 0 & 0 & (K_3)_{4,3} & (K_3)_{4,4} \end{bmatrix}$$

Add known nodal forces

$$P = \begin{pmatrix} 50 \\ -100 \\ P_3 \\ P_4 \\ P_5 \\ P_6 \\ P_7 \\ P_8 \end{pmatrix} \cdot \text{kN} \quad \text{Therefore} \quad \begin{aligned} P_1 &:= 50 \cdot \text{kN} \\ P_2 &:= -100 \cdot \text{kN} \end{aligned}$$

Assemble non-nodal force matrices.

Members 1 and 2 have no non-nodal forces; member 3 has a temperature load.

In local coordinates:

$$Q_{F_1} := E_1 \cdot \alpha_1 \cdot A_1 \cdot \Delta T \cdot \begin{pmatrix} 1 \\ -1 \end{pmatrix} \quad Q_{F_1} = \begin{pmatrix} 324 \\ -324 \end{pmatrix} \text{ kN}$$

$$Q_{F_2} := \begin{pmatrix} 0 \\ 0 \end{pmatrix} \cdot \text{kN} \quad Q_{F_3} := \begin{pmatrix} 0 \\ 0 \end{pmatrix} \cdot \text{kN}$$

Transform non-nodal forces to global coordinates

$$F_{F_i} := T_i^T \cdot Q_{F_i} \quad F_{F_1} = \begin{pmatrix} -229.103 \\ 229.103 \\ 229.103 \\ -229.103 \end{pmatrix} \text{ kN} \quad F_{F_2} = \begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \end{pmatrix} \text{ kN} \quad F_{F_3} = \begin{pmatrix} 0 \\ 0 \\ 0 \\ 0 \end{pmatrix} \text{ kN}$$

Assemble non-nodal force matrix

$$P_F := \begin{bmatrix} (F_{F_1})_1 + (F_{F_2})_1 + (F_{F_3})_1 \\ (F_{F_1})_2 + (F_{F_2})_2 + (F_{F_3})_2 \\ (F_{F_1})_3 \\ (F_{F_1})_4 \\ (F_{F_2})_3 \\ (F_{F_2})_4 \\ (F_{F_3})_3 \\ (F_{F_3})_4 \end{bmatrix} \quad P_F = \begin{pmatrix} -229.103 \\ 229.103 \\ 229.103 \\ -229.103 \\ 0 \\ 0 \\ 0 \\ 0 \end{pmatrix} \text{ kN}$$

Displacement degrees of freedom 3-8 are supported.

Extract the reduced stiffness matrix.

$$S_{\text{red}} := \text{submatrix}(S, 1, 2, 1, 2) \quad S_{\text{red}} = \begin{pmatrix} 1.3 \times 10^5 & -16605.3 \\ -16605.3 & 46180.7 \end{pmatrix} \frac{\text{kN}}{\text{m}}$$

Extract the reduced nodal force matrix

$$P_{\text{red}} := \begin{pmatrix} P_1 \\ P_2 \end{pmatrix} \quad P_{\text{red}} = \begin{pmatrix} 50 \\ -100 \end{pmatrix} \text{ kN}$$

Extract the reduced non-nodal force matrix.

$$P_{\text{Fred}} := \begin{pmatrix} P_{F_1} \\ P_{F_2} \end{pmatrix} \quad P_{\text{Fred}} = \begin{pmatrix} -229.103 \\ 229.103 \end{pmatrix} \text{ kN}$$

Calculate displacements

$$d_{\text{red}} := S_{\text{red}}^{-1} \cdot (P_{\text{red}} - P_{\text{Fred}}) \quad d_{\text{red}} = \begin{pmatrix} 1.251 \\ -6.677 \end{pmatrix} \text{ mm}$$

Reassemble structural displacement vector.

$$d := \begin{pmatrix} d_{\text{red}_1} \\ d_{\text{red}_2} \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \\ 0 \end{pmatrix} \quad P := S \cdot d + P_F \quad P = \begin{pmatrix} 50 \\ -100 \\ -51.176 \\ 51.176 \\ -83.389 \\ 0 \\ 84.565 \\ 48.824 \end{pmatrix} \text{ kN}$$

Calculate the support reactions

Member displacements

For element 1

Element 2

$$v_1 := \begin{pmatrix} d_1 \\ d_2 \\ d_3 \\ d_4 \end{pmatrix} \quad v_1 = \begin{pmatrix} 1.251 \\ -6.677 \\ 0 \\ 0 \end{pmatrix} \text{ mm} \quad v_2 := \begin{pmatrix} d_1 \\ d_2 \\ d_5 \\ d_6 \end{pmatrix} \quad v_3 := \begin{pmatrix} d_1 \\ d_2 \\ d_7 \\ d_8 \end{pmatrix}$$

Member forces in local coordinates

$$Q_i := k_i \cdot T_i \cdot v_i + Q_{F_i}$$

$$Q_1 = \begin{pmatrix} -72.374 \\ 72.374 \end{pmatrix} \text{ kN}$$

$$Q_2 = \begin{pmatrix} -83.389 \\ 83.389 \end{pmatrix} \text{ kN}$$

$$Q_3 = \begin{pmatrix} 97.647 \\ -97.647 \end{pmatrix} \text{ kN}$$